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The Unmanned Aerial Vehicle Routing and Trajectory Optimisation Problem, a taxonomic review

Abstract

Over the past few years, Unmanned Aerial Vehicles (UAVs) have become more and more popular. The complexity of routing UAVs has not been fully investigated in the literature. In this paper, we provide a formal definition of the UAV Routing and Trajectory Optimisation Problem (UAVRTOP). Next, we introduce a taxonomy and review recent contributions in UAV trajectory optimisation, UAV routing and articles addressing these problems, and their variants, simultaneously. We conclude with the identification of future research opportunities.

Keywords: unmanned aerial vehicles, routing, trajectory optimisation, literature review, taxonomy

1. Introduction

Unmanned Aerial Vehicles (UAVs) are aircraft that do not need a human pilot on board. In general, these vehicles are either controlled by an embedded computer or by a pilot operating a remote control. Drones, remote controlled helicopters and unmanned gliders are examples of UAVs. Gliders differ from the other types due to the lack of on-board propulsion (e.g., an electric or combustion engine). Modern UAVs were first developed in the 1920s to support military operations in which the presence of human pilots was either impossible or too dangerous (Beard & McLain, 2012; Keane & Carr, 2013). However, UAVs have recently become very popular for logistics and surveillance applications (Tsourdos et al., 2010).

A report from the National Purchase Diary has shown that sales of drones increased by 224% in twelve months from April 2015, reaching a total of 200 million dollars (NPD, 2016). Due to being able to embed several transmitters, sensors and photographic equipment, UAVs can be used in a large range of applications. Successful cases have been reported in, for example, aerial reconnaissance (Ruzgiené et al., 2015), aerial forest fire detection (Yuan et al., 2015), target observation (Rysdyk, 2006), traffic monitoring and management (Kanistras et al., 2013), online commerce (Wang et al., 2017), geographical monitoring (Uysal et al., 2015), scientific data collection (Stöcker et al., 2015), meteorological sampling (Elston et al., 2014) and disaster assessment and response (Quaritsch et al., 2010; Xu et al., 2014; Nedjati et al., 2016). In Hayat et al. (2016), several applications of UAV networks are reviewed. The use of UAVs for 3D mapping is surveyed in Nex & Remondino (2013). A literature review about the applications of UAVs in humanitarian relief is provided by Bravo & Leiras (2015). More examples of the growing applications of UAVs are presented in Rao et al. (2016).

The academic routing community has acknowledged the interest of companies and organisations in adopting UAVs in their operations. A recent example is the approach of combining UAVs and trucks for distribution activities by dispatching drones from trucks for the last mile distribution within city centres

(Ha et al., 2015; Murray & Chu, 2015; Wang et al., 2017). It has been shown that this solution can reduce truck travel time, and the corresponding CO_2 emissions, by up to 50%. The UAV Task Assignment Problem (UAVTAP), which is closely related to the UAV routing problem, consists of optimising the assignment of a set of UAVs to a set of tasks subject to mission constraints (Khamis et al., 2015). A growing body of literature appeared on the UAVTAP in the last decade, e.g., Ramirez-Atencia et al. (2016), Wang et al. (2015), Hu et al. (2015a), Thi et al. (2012), Alidaee et al. (2010) and Edison & Shima (2011). However, the UAV routing and task assignment literatures have often neglected constraints due to the flight dynamics of the UAVs. Finding feasible trajectories for UAVs in a routing problem is a complex task, but it is necessary to ensure the feasibility of the UAVs routes. For some real-world applications involving more complex UAV systems, such as unmanned gliders and fixed-wing vehicles, the definition of routes must be coupled to the design of flyable trajectories, otherwise the assigned routes might become inefficient or even infeasible for these UAVs.

Most of the UAVs used for civil applications present a low flight autonomy. Therefore, it is important for UAV routing algorithms to properly model battery life. According to Fügenschuh & Müllenstedt (2015), this can be achieved by integrating the UAVs' dynamics with routing. As mentioned by the authors, for powered UAVs, a proper modelling of the actual fuel consumption must include, for instance, the current weight, the altitude, the speed and climb/descent rate, which are usually modelled by flight dynamics.

Zhang et al. (2012) consider a problem where a UAV must visit a set of targets. However, after reaching a predetermined distance from a target the UAV must then adjust its *flight attitude* (i.e., its orientation) in order to perform a payload delivery. After the delivery, the UAV must complete an *escape manoeuvre* and prepare for the next delivery. According to Zhang et al. (2012), routing and trajectory optimisation must be integrated in order to ensure the safety of the vehicle and the feasibility of trajectories.

The computation of trajectories for UAVs has been widely studied in the aerospace engineering and optimal control literature (Yang et al., 2016). The Trajectory Optimisation (TO) problem consists of finding a *control history* of a given vehicle, that minimises a scalar performance index (for example, flight time or fuel consumption) while satisfying constraints on the kinematics (position, velocity and acceleration) and the dynamics (forces and moments) of the vehicle (Betts, 1998). A trajectory is generally associated with a set of Equations of Motion (EOMs) that describe the relationship between the spatial and the temporal changes to the system. The TO problem is closely related to the Optimal Control (OC) problem (Betts, 2001).

The problem named Path Planning (PP) consists of finding a flyable path for a UAV visiting a given sequence of waypoints (targets) in a two-dimensional (2D) or three-dimensional (3D) space without considering the vehicle's dynamics. According to Gasparetto et al. (2015), PP is a *geometric* problem, because it is defined as finding a geometric path regardless any specified time law. In turn, TO consists of assigning a time law to a controlled geometric path.

More complex variants of the PP problem including, for instance, wind and motion constraints, require substantial simplifications and assumptions to be solved heuristically (Kunchev et al., 2006; Rathinam & Sengupta, 2007). The books by Tsourdos et al. (2010) and Beard & McLain (2012) provide good overviews of PP algorithms for UAVs. On the other hand, high fidelity TO models (i.e., using more

accurate physical models) have been developed for aircraft and spacecraft (Raivio et al., 1996; Conway, 2010; Fisch, 2011; García-Heras et al., 2014; Colasurdo et al., 2014). These models are currently solved by OC techniques. An overview of OC methods for TO is provided in Betts (1998, 2001).

The field of TO has however not considered routing decisions: given a set of ordered waypoints, it is possible to find a feasible trajectory for a generic UAV, but it is not clear in the literature if the sequence of waypoints is appropriate. For example, for a gliding vehicle (i.e., with no onboard thrust) a given waypoint sequence might be infeasible in terms of flight dynamics. Given a fleet of UAVs, it is an open question how to combine routing and trajectory decisions in a single optimisation problem. As far as the authors are aware, there is not a survey summarising the literature about routing and trajectory optimisation for UAVs.

Research about integrated routing and TO problems seems to be still fragmented. One of the main contributions of this paper is introducing the UAV Routing and Trajectory Optimisation Problem (UAVR-TOP). We believe that integrating TO and routing in a single optimisation problem is a key research challenge in adopting UAVs for real world applications.

The purpose of this survey is to present the UAVRTOP, highlighting approaches already proposed in the literature and providing a direction for further research. We introduce a taxonomy, that is able to identify the key components of routing and TO problems, as well as highlight assumptions and simplifications commonly adopted in the literature.

The remainder of this paper is organised as follows. In Section 2, we formally define the UAVRTOP. In Section 3, a background on TO problems is provided. The same is done in Section 4 for vehicle routing problems. In Section 5, a taxonomy of UAV routing and TO problems is provided. An application of the proposed taxonomy to a selected number of papers is demonstrated in Section 6. This section continues with an analysis of the results obtained from the taxonomic review. In Section 7, we discuss future research opportunities.

2. The UAV routing and trajectory optimisation problem

In this section, we formally define the UAV Routing and Trajectory Optimisation Problem (UAVR-TOP), the problem in which a fleet of UAVs has to visit a set of waypoints assuming generic kinematics and dynamics constraints. Wind conditions, collision avoidance between UAVs and obstacles can also be incorporated in the model.

2.1. A mathematical formulation for the UAVRTOP

In the following, we assume a fleet C of UAVs is available at the launching site 0. Let $G = (V, A)$ be a graph, where the set V represents all the waypoints that need to be visited by the UAVs and A represents the set of arcs between waypoints. In addition, let $0'$ represent the landing site. The cost of using a vehicle $k \in C$ is F_k. The parameters (e.g., mass, wing area, aerodynamics coefficients) of the UAV k travelling between i and j are stored in the vector $\mathbf{p}_{ijk}$. Note that these parameters may change during the mission due, for example, to a change in flight mode (if hybrid UAVs are used). The state of a UAV is a vector fully defining the position, orientation and velocity of the vehicle in some coordinate system (alternative state representations will be described in Section 3).

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For simplicity, we recall $\mathbf{y}_{ijk}(t_{ijk}) \in \mathbb{R}^{n_y^k}$, $n_y^k \in \mathbb{Z}$, the state variable of the UAV k travelling between waypoints i and j at time $t_{ijk} \in \mathbb{R}$. Similarly, the control variables model the inputs that are given to the physical systems in order to achieve a desired trajectory. Typical control variables for UAVs are the thrust (the impulse given by the UAV engine, if any), the roll angle, a.k.a. bank angle (which banks the aircraft to change its horizontal flight direction), and the angle-of-attack (which is related to how much lift the aircraft's wing generate). We define $\mathbf{u}_{ijk}(t_{ijk}) \in \mathbb{R}^{n_u^k}$, $n_u^k \in \mathbb{Z}$, the control variables for a UAV k flying on arc (i,j) at time $t_{ijk} \in \mathbb{R}$.

The physical laws governing the UAV k travelling between the waypoints i and j at time t_{ijk} are referred as *system dynamics*. In general terms, the system dynamics can be expressed by a set of EOMs in the form of a system of Ordinary Differential Equations (ODEs) as follows:

$$\dot{\mathbf{y}}_{ijk} = \mathbf{f}_k(\mathbf{y}_{ijk}(t_{ijk}), \mathbf{u}_{ijk}(t_{ijk}), \mathbf{p}_{ijk}, t_{ijk}) \forall i,j \in V, \forall k \in C \tag{1}$$

The functions $\mathbf{f}_k, \forall k \in C$, in the right hand side of the EOMs (1), represent the relationship between the variables and parameters with the derivatives over time of the state variables (here denoted by " ˙ ").

State and control variables have to be specified for a time instant to initialise the ODEs. In what follows, we assume that the initial conditions need to be specified at time $t = 0$. It is also reasonable to assume that only the control variables need to be optimised since the values of the states can be determined, provided an initial condition and the evolution of the controls over time.

Let $w_k(.)$ be a function computing the cost of using UAV k along an arbitrary trajectory. The routing cost for a UAV k to travel between waypoints i and j can be computed as:

$$\int_{t_{ijk}^o}^{t_{ijk}^f} w_k(\mathbf{y}_{ijk}(t_{ijk}), \mathbf{u}_{ijk}(t_{ijk}), \mathbf{p}_{ijk}, t_{ijk}) \mathrm{d}t_{ijk}. \tag{2}$$

The variables t_{ijk}^o and t_{ijk}^f represent the initial and final flight times of the UAV k travelling between waypoints i and j such that $t_{ijk} \in [t_{ijk}^o, t_{ijk}^f]$.

Bounds on the state and control variables are usually imposed by a given UAV technology. We denote $\mathbf{y}_{ijk}^{lb}$ and $\mathbf{y}_{ijk}^{ub}$ the lower and upper bounds on the state variables $\mathbf{y}_{ijk}(t_{ijk})$ of the UAV k travelling in an arc (i,j) for all $t_{ijk} \in \mathbb{R}$, respectively. Similarly, $\mathbf{u}_{ijk}^{lb}$ and $\mathbf{u}_{ijk}^{ub}$ represent the lower and upper bounds of the control variables $\mathbf{u}_{ijk}(t_{ijk})$ of the UAV k travelling on arc (i,j) for all $t_{ijk} \in \mathbb{R}$. We also assume lower and upper bounds on the operational constraints, here denoted as $\mathbf{g}_{ijk}^{lb}$ and $\mathbf{g}_{ijk}^{ub}$.

According to our assumption on the initial conditions, the initial flight time from the launching point must be defined as $t_{0jk}^o = 0, \forall j \in V, \forall k \in C$. Let $\bar{\mathbf{y}}_o$ and $\bar{\mathbf{u}}_o$ represent predetermined initial conditions. Thus, the initial state and control variables can be defined as $\mathbf{y}_{0jk}(t_{0jk}^o) = \bar{\mathbf{y}}_o$ and $\mathbf{u}_{0jk}(t_{0jk}^o) = \bar{\mathbf{u}}_o$, respectively, if UAV k departs from the launching point.

Let us define the following binary variables:

$$x_{ijk} = \begin{cases} 1, & \text{if UAV } k \text{ flies directly from waypoint } i \text{ to } j \\ 0, & \text{otherwise.} \end{cases} \tag{3}$$

Hereafter, we will describe the optimisation problem defined by Equations (4-19). This formulation is a conceptual model created for describing the UAVRTOP in mathematical terms. The objective function (4) minimises the sum of the fixed cost of using a UAV, the routing cost of flying between waypoints i

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and j and a measure of the quality of the trajectories at the end points of each arc (i,j). Non desirable features at the end points of the UAVs' trajectories can be penalised in the objective function by means of the functions $\phi_k(\mathbf{y}_{ijk}(t_{ijk}^f), \mathbf{u}_{ijk}(t_{ijk}^f), \mathbf{p}^{ijk}, t_{ijk}^f)$. Such undesirable characteristics may include, e.g., sharp flight angles, prohibited flight speeds and noise levels (Vanderbei, 2001; Zhang et al., 2012). Constraints (5) and (6) ensure that every waypoint is visited exactly once and that, if a UAV arrives at a waypoint $l \in V$, it must also depart from l. Constraints (7) make sure that each UAV departs from the launching point 0 and lands in $0'$, if the UAV k is used. Constraints (8) ensure that the UAVs' dynamics are preserved if arc (i,j) is used in a solution. In a similar way, Constraints (9-11) make sure the bounds on the state variables, control variables and *operational constraints* $(\mathbf{g}_{ijk}(\mathbf{y}_{ijk}(t_{ijk}), \mathbf{u}_{ijk}(t_{ijk}), \mathbf{p}_{ijk}, t_{ijk}))$ are respected for every arc (i,j) and for every UAV k if these are travelled in the obtained solution. These constraints can model, for example, collision avoidance and undesirable manoeuvres. Constraints (12) and (13) ensure that the final state and control variables at every arc (i,j) visited by UAV k is linked to the state and control variables of its subsequent arc (j,l) if waypoints i, j and l are visited by UAV k in this order. Constraints (14) preserve the continuity of the time variable $t_{ijk}, \forall i, j \in V$, along the UAV's k trajectory for all $k \in C$. Constraints (15) and (16) provide the initial states and controls for every UAV departing from the launching point. Finally, Constraints (17-19) define the domain of the variables.

The UAVRTOP can be modelled as follows:

$$\min \quad \sum_{k \in C} \sum_{i \in V} F_k x_{0ik}$$

$$+ \sum_{k \in C} \sum_{(i,j) \in A} \left\{ \int_{t_{ijk}^o}^{t_{ijk}^f} w_k(\mathbf{y}_{ijk}(t_{ijk}), \mathbf{u}_{ijk}(t_{ijk}), \mathbf{p}_{ijk}, t_{ijk}) \mathrm{d}t_{ijk} \right\} x_{ijk}$$

$$+ \sum_{k \in C} \sum_{(i,j) \in A} \phi_k(\mathbf{y}_{ijk}(t_{ijk}^f), \mathbf{u}_{ijk}(t_{ijk}^f), \mathbf{p}^{ijk}, t_{ijk}^f) x_{ijk} \tag{4}$$

$$\text{s.t.} \quad \sum_{k \in C} \sum_{\substack{i \in V \cup \{0\} \\ i \neq j}} x_{ijk} = 1, \forall j \in V \tag{5}$$

$$\sum_{\substack{i \in V \cup \{0\} \\ i \neq j}} x_{ijk} - \sum_{\substack{i \in V \cup \{0'\} \\ i \neq j}} x_{jik} = 0, \forall j \in V, \forall k \in C \tag{6}$$

$$\sum_{i \in V} x_{0ik} = \sum_{i \in V} x_{i0'k} \leq 1, \forall k \in C \tag{7}$$

$$\dot{\mathbf{y}}_{ijk} = \mathbf{f}_k(\mathbf{y}_{ijk}(t_{ijk}), \mathbf{u}_{ijk}(t_{ijk}), \mathbf{p}_{ijk}, t_{ijk}) x_{ijk}, \forall i, j \in V, \forall k \in C \tag{8}$$

$$\mathbf{g}_{ijk}^{lb} x_{ijk} \leq \mathbf{g}_{ijk}(\mathbf{y}_{ijk}(t_{ijk}), \mathbf{u}_{ijk}(t_{ijk}), \mathbf{p}_{ijk}, t_{ijk}) \leq \mathbf{g}_{ijk}^{ub} x_{ijk},$$

$$\forall i, j \in V, \forall k \in C \tag{9}$$

$$\mathbf{y}_{ijk}^{lb} x_{ijk} \leq \mathbf{y}_{ijk}(t_{ijk}) \leq \mathbf{y}_{ijk}^{ub} x_{ijk}, \forall i, j \in V, \forall k \in C \tag{10}$$

$$\mathbf{u}_{ijk}^{lb} x_{ijk} \leq \mathbf{u}_{ijk}(t_{ijk}) \leq \mathbf{u}_{ijk}^{ub} x_{ijk}, \forall i, j \in V, \forall k \in C \tag{11}$$

$$\mathbf{y}_{jlk}(t_{jlk}^o) x_{ijk} = \mathbf{y}_{ijk}(t_{ijk}^f) x_{ijk} x_{jlk}, \forall i, j, l \in V, \forall k \in C \tag{12}$$

$$\mathbf{u}_{jlk}(t_{jlk}^o) x_{ijk} = \mathbf{u}_{ijk}(t_{ijk}^f) x_{ijk} x_{jlk}, \forall i, j, l \in V, \forall k \in C \tag{13}$$

$$t_{jlk}^o x_{ijk} = t_{ijk}^f x_{ijk} x_{jlk}, \forall i, j, l \in V, \forall k \in C \tag{14}$$

$$\mathbf{y}_{0jk}(t_{0jk}^o) = \bar{\mathbf{y}}_o x_{0jk}, \forall j \in V, \forall k \in C \tag{15}$$

$$\mathbf{u}_{0jk}(t_{0jk}^o) = \bar{\mathbf{u}}_o x_{0jk}, \forall j \in V, \forall k \in C \tag{16}$$

$$x_{ijk} \in \{0,1\}, \forall i,j \in V, \forall k \in C \tag{17}$$

$$\mathbf{u}_{ijk}(t_{ijk}) \in \mathbb{R}^{n_u^k}, \forall i,j \in V, \forall k \in C \tag{18}$$

$$t_{ijk}, t_{ijk}^o, t_{ijk}^f \in \mathbb{R}. \forall i,j \in V, \forall k \in C \tag{19}$$

3. The trajectory optimisation problem

TOPs are a special case of OC problems determining the trajectory of a system (e.g., vehicles such as spacecraft, aircraft, UAVs) while minimising a measure of performance and satisfying a set of boundary (initial and final) conditions, path constraints and system dynamics.

The origin of OC problems dates to as early as the 17th century when Johann Bernoulli proposed the Brachistochrone problem (Ross, 2009), one of the first problems in *calculus of variations*. One of the first applications of the calculus of variations to the control of flying vehicles was presented by Robert Goddard in "A method of reaching extreme altitudes" (Goddard, 1919), where the objective was to determine the minimum initial mass of a ground-based rocket necessary to achieve a given altitude. OC methods are a classical tool in the computation of spacecraft trajectories, e.g., for interplanetary travel and satellite transfer orbits around Earth (Conway, 2010; Colasurdo et al., 2014).

Usually, system dynamics are modelled by a set of EOMs that can be nonlinear and discontinuous. *six degrees of freedom* (6DOF) EOMs are composed by translational equations (containing forces, position, velocity, acceleration, etc.) and rotational equations (containing moments, angular velocities, angular acceleration, etc.). Under simplifying assumptions, 6DOF EOMs can be decoupled into *three degrees of freedom* (3DOF) EOMs, see, e.g., Stengel (2004) and Fisch (2011). Usual state variables in 6DOF EOMs are the position vector, velocity, pitch angle, pitch rate, weight and flight path angle. In the 3DOF case, the state vector can represent, for instance, the position, velocity, flight path angle and yaw angle of the vehicle.

Solving a Trajectory Optimisation Problem (TOP) for an aircraft consists of generating the inputs for the aircraft's control system so as to perform a desired set of manoeuvres. A TOP takes as input the dynamic constraints of the aircraft and outputs time-indexed states and controls such as positions, velocities and accelerations.

Other difficulties can be added to the problem if one considers that the boundary conditions depend on unknown variables or if the dynamics of the vehicles change over time. In this cases, TOPs can be divided into two or more *phases* in order to properly model the changes in the operational or physical characteristics of the vehicles. A phase can be defined as a segment of a trajectory in which the dynamical system remains unchanged. Phases can be described by their own boundary conditions, system of differential equations, operational constraints and time events. Finally, all phases can be linked or not depending on the behaviour of the dynamical system.

Aircraft TO models have gained much popularity over the last decades. For instance, Schultz & Zagalsky (1972) present solutions for several fixed endpoint aircraft TOPs using calculus of variations. In Raivio et al. (1996), a nonlinear programming-based method is proposed to compute optimal trajectories for a descending aircraft. Fisch (2011) presents a high fidelity optimisation framework for the computation

of air race trajectories under safety requirements. García-Heras et al. (2014) compare several OC methods for the TO of cruise flight with fixed arrival time. Finally, Delahaye et al. (2014) present a survey of mathematical models for the computation of aircraft trajectories.

OC methods for UAVs are similar to those of full size aircraft, and therefore similar system dynamics can be used for both types of planes. On the other hand, new challenges are introduced when specific mission demands are requirements. Moreover, due to their limited capacity, extra effort must be put into determining successful flight plans. Therefore, algorithms that are capable of tackling the UAVs' particularities while developing flight plans must be developed.

3.1. Direct and indirect methods for trajectory optimisation problems

Two main classes of numerical methods became very popular for solving TOPs, these being, *direct* and *indirect* methods. The so-called *direct* methods rely on the discretisation of a infinite-dimensional OC problem into a finite-dimensional optimisation problem. This strategy is commonly known as "discretise, then optimise". In a *direct single shooting* method, for example, the controls are discretised on a fixed grid using an arbitrary parametrisation scheme. The next step of this method consists of solving a non-linear programming problem in order to find an optimal vector of parameters. The *indirect* methods consist of determining necessary optimality conditions for an OC problem and then using a discretisation method to solve the resulting equations. Indirect methods generally apply an "optimise, then discretise" strategy. In an *indirect single shooting* method, for example, the resulting optimality conditions consist of a boundary value problem, which can be solved by means of a simple single shooting algorithm (Betts, 2001).

Several sophisticated algorithms have been developed for solving TOPs. Reviewing such works is considered beyond the scope of this paper. More information about algorithms for OC and TO can be found, for example, in the papers by Stryk & Bulirsch (1992), Betts (1998), Ross (2009), Wang (2009) and Rao (2014); and, the books by Bryson (1975), Bertsekas (1979), Betts (2001), Bryson (2002) and Kirk (2012).

3.2. The UAV path planning problem

Using the notation defined by Latombe (1991), the basic PP problem can be defined as follows. Let $\mathcal{A}$ be an object (a robot) moving in a workspace $\mathcal{S}$ (e.g., in an Euclidean space $\mathcal{S} = \mathbb{R}^n, n = 2$ or 3). A set of obstacles $\mathcal{B}_1, \ldots, \mathcal{B}_m$ is assumed to be distributed over $\mathcal{S}$. The problem consists in, given initial and final *configurations* (position and orientation) for $\mathcal{A}$, find a path in $\mathcal{S}$ that avoids collisions with the objects $\mathcal{B}_1, \ldots, \mathcal{B}_m$. It has been shown that this problem is $\mathcal{NP}$-hard if the velocity of the object $\mathcal{A}$ is unbounded and no rotation is considered (Reif & Sharir, 1994). For Gasparetto et al. (2015), a path planning problem consists of finding a collision-free path among an environment from an initial point to a final goal. For example, YongBo et al. (2017) studied a path planning problem with obstacles in three dimensions. In the literature, the terms PP and *motion planning* are used almost interchangeably (Barraquand & Latombe, 1991). Both problems have gained much popularity over the years. Figure 1 shows the number of publications by year on UAV PP problems.

PP algorithms can be classified into discrete and continuous methods. In the former, the workspace $\mathcal{S}$ is transformed into a graph through discretisation. Conventional heuristics or exact shortest path
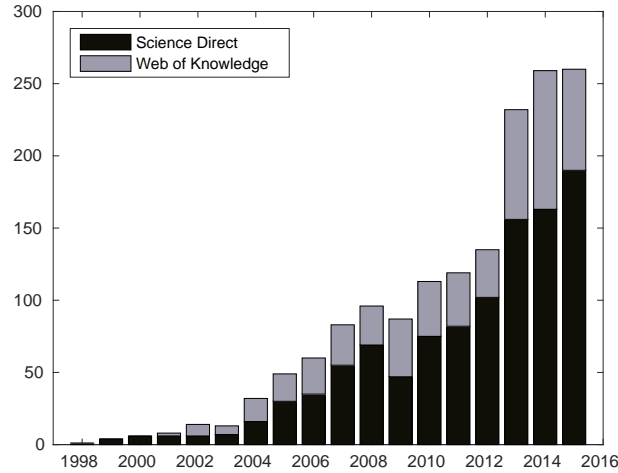
Figure 1: Number of published papers by year on PP problems.

algorithms are then used to find a path between a given initial configuration and a final configuration. The output for discrete methods are usually polygonal paths, i.e., paths with no curvature constraints. Therefore, in the case of UAVs these paths need to be further refined. Continuous methods represent $\mathcal{S}$ by using a continuous function. Tsourdos et al. (2010), for instance, employed attraction fields to represent the desired endpoints and repulsive fields to represent obstacles in order to produce a collision-free UAV path.

Problems integrating UAV routing and PP have been studied before, see, for example, Manyam et al. (2015), Ho & Ouaknine (2015), Enright et al. (2015), Sundar & Rathinam (2014) and Levy et al. (2014). Under simplifying assumptions, a PP problem can be modelled as a network problem and standard shortest path techniques can be used. A common assumption is that the UAV can be modelled as a Dubin's vehicle (Medeiros & Urrutia, 2010). A Dubin's vehicle has a limited turning angle and is restricted to move forward, therefore it can be a good representation for some types of UAVs. This simplification is very popular specially for modelling rotary-wing aircraft such as quadcopters. However, for most fixed-wing UAVs, the Dubin's assumption might not be suitable due to their complicated dynamics. The reader is referred to Tsourdos et al. (2010) for more details on UAVs PP methods.

Most algorithms for UAV PP have originated from adaptations of existing algorithms for robot PP. However, we do not intend to survey all PP algorithms as it has already been done in other articles, e.g., Kunchev et al. (2006), Goerzen et al. (2009), Galceran & Carreras (2013) and Yang et al. (2016).

4. The vehicle routing problem

The Vehicle Routing Problem (VRP) is a very well known problem in operational research and combinatorial optimisation. In the VRP, routes must be assigned to a set of vehicles that must serve a set of customers such that the total cost of the operation is minimised. Its classical variant is called Capacitated Vehicle Routing Problem (CVRP), where a load capacity is assigned to each vehicle.

The CVRP can be formally defined as follows. A set of vertices $V = \{0, \ldots, n\}$ and a set of arcs A connecting these vertices are given. Each vertex represents a customer with demand d_i, $i \in V \setminus \{0\}$. A value c_{ij} is assigned to each arc $(i, j) \in A$ representing the travel cost between two customers. Let $C = \{1, \ldots, m\}$ be a set of homogeneous vehicles with capacity Q. Here we denote the vertex $i = 0$

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representing the depot (launching site). The CVRP consists of finding a minimum cost set of m routes starting and ending at the depot such that all customers are visited exactly once, all customers' demands are satisfied and the capacity of the vehicles are respected. The CVRP is well known to be $\mathcal{NP}$-hard. More information about the VRP and its variants can be found, e.g., in Golden & Assad (1988), Cordeau et al. (2007), Golden et al. (2008), Toth & Vigo (2002), Eksioglu et al. (2009), Lahyani et al. (2015) and Braekers et al. (2016).

The m-TSP is closely related to the VRP. In the m-TSP, m minimum cost tours starting at the depot must be found such that every vertex in $V \setminus \{0\}$ is visited exactly once. The m-TSP can be reduced to the CVRP if all vehicles are considered to have infinite capacity. An extensive literature review on models and algorithms for the m-TSP is presented by Bektas (2006).

The VRP and the m-TSP have been widely studied for terrestrial applications, but with the development of new technologies, such as unmanned vehicles, new variants of these problems are gaining interest among the scientific community. The problem of routing an aerial vehicle is more complex than the VRP because it combines the combinatorial characteristics of the VRP with the complexity of dealing with the system dynamics of UAVs (i.e., flight dynamics, battery life, wind conditions).

4.1. UAV task assignment problem

The UAV Task Assignment Problem (UAVTAP) consists of finding an optimal assignment of UAVs to a set of tasks. Often, the UAVs have different characteristics and the tasks depend on the nature of the application. It has been shown that this problem is $\mathcal{NP}$-hard (Alidaee et al., 2010). Due to the quick development of UAV technology, new challenging assignment problems arise every day and many algorithms have been developed to address the new challenges. Figure 2 shows the number of publications by year in UAVTAPs, the y axis corresponds to the number of publications found in Science Direct and Web of Knowledge databases. One can observe that this field of research has gained attention of the scientific community. A detailed literature review about algorithms for multi-robot Task Assignment (TA) problems can be found in Khamis et al. (2015).
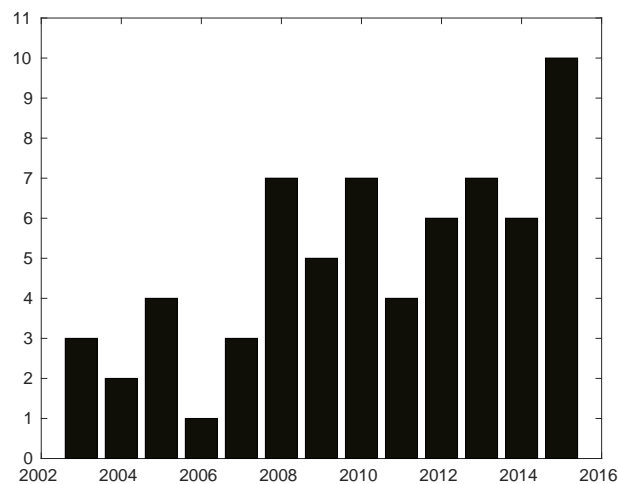

Figure 2: Number of published papers by year on TA problems.

The UAVTAP shares some characteristics with the VRP. Many examples in the literature support this claim. One can cite, for instance, TA with time windows (Karaman & Inalhan, 2008), multi-depot

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(launching points) (Darrah et al., 2012), task allocation with resource constraints (Kim et al., 2015), TA with flexible demand (Alidaee et al., 2011), real-time and dynamic assignment (Kim et al., 2007; Lin et al., 2013), time-dependent TA (Kingston & Schumacher, 2005), and finally, TA under uncertainty (Alighanbari & How, 2008; Hu et al., 2015a). However, the UAVTAP differs from the VRP by allowing, for example, multiple visits and subtours. In addition, new features may also be introduced, e.g., the possibility of heterogeneous UAVs to perform multiple operations at the same time (Shima & Schumacher, 2009). On the other hand, the kinematics and dynamics of UAVs are usually not considered, as opposed to the formulation of path/motion planning and TO problems.

5. UAVRTOP taxonomy

In this section, a taxonomy is proposed in order to help readers identify the key differences among various UAV routing problems and guide their research towards the development of new algorithms.

We have identified 20 attributes that are common to the UAV TO, PP, routing and TA literatures. They define common features of UAVs routing problems, such as the kind of fleet, mission characteristics and flight dynamics. Attributes are further grouped into five classes. The first class collects the characteristics of UAVs, the second class represents the characteristics of waypoints, the third class describes the characteristics of the environment, the fourth class involves the characteristics of the launching point(s) and the last class is concerned about flight duration. These are listed in Table 1. The last two lines of this table are not part of the taxonomy, but they are important for understanding Appendix A.

Within class **UAVs**, the fleet and how the UAVs' kinematics and dynamics are modelled are considered as follows:

1. *Multiple* - A fleet of vehicles is available (as opposed to a single vehicle).
2. *Heterogeneous* - Heterogeneous fleet, i.e., the fleet is composed of vehicles with different characteristics (as opposed to a homogeneous fleet).
3. *Fleet Size* - The size of the fleet must be optimised (as opposed to a fixed fleet size).
4. *Capacity* - Vehicles are capacitated (as opposed to uncapacitated vehicles). A capacitated UAV might have, for example, a maximum flight endurance or maximum payload capacity.
5. *Geometric* - The vehicle's flight dynamics are neglected (as opposed to considering flight dynamics). For example, by using Euclidean distances between waypoints instead of flight distances.
6. *Dubin's* - Dubin's vehicles are used to model the UAVs.
7. *EOM* - A set of differential equations has been used to model the vehicles' kinematics and dynamics (as opposed to neglecting the dynamics).

The class **Waypoints** presents the attributes of the waypoints (vertices):

8. *Multiple* - Multiple waypoints must be visited (as opposed to a single waypoint or destination).
9. *Unordered* - The visiting order of waypoints is unknown (as opposed to a predefined order).
10. *Visits* - The waypoints can be visited multiple times (as opposed to a single visit for each waypoint).
11. *Constraints* - Special mission constraints must be considered. For instance, time-windows, precedences, and special boundary conditions.

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12. *Covering Region* - A continuous, but not necessarily convex, region (or airspace) is defined over the waypoint. We believe this characteristic is important to the UAVRTOP since most UAVs' sensors require at least a minimum radius of action in order to be effective.

The **Environment** class collects the attributes about the environment where the UAVs operate:

13. *3D* - The UAVs operate in a 3D space (as opposed to a 2D space).

14. *Obstacles* - The problem includes the presence of fixed or moving obstacles (as opposed to an obstacle-free environment).

15. *Wind* - The effects of wind are considered (as opposed to neglecting the wind effects).

16. *Real-time* - The problem must be solved in real-time. For example, waypoints and tasks arriving at random times and locations.

The class **Launching** groups the attributes about the number of launching points (depots):

17. *Multiple* - There are multiple launching points (as opposed to a single launching point).

18. *Inter-depot* - There are inter-depots available (e.g., for refuelling, battery replenishment or maintenance of the UAVs).

Papers are classified in class **Time** according to the way the flight time is considered in their models:

19. *Fixed* - The UAVs' flight times between arcs can be computed beforehand. This is a common characteristic of some PP methods (e.g., the Dubin's model).

20. *Variable* - The UAVs' velocities and flight times between arcs are optimisation variables.

In order to provide a survey of the most relevant and recent papers, we adopted the following procedure. Papers published since 2010 were collected from the following databases: *The Web of Science, Google Scholar* and *ScienceDirect*. We have limited our search to papers published in English. In order to cover the most common types of UAVs, we considered Unmanned Combat Aerial Vehicle (UCAV), Unmanned Aerial Systems (UAS) and *aerial gliders* in our search. The following keywords were used:

- UAV/UCAV/UAS/aerial glider trajectory optimisation

- UAV/UCAV/UAS/aerial glider PP

- UAV/UCAV/UAS/aerial glider motion planning

- UAV/UCAV/UAS/aerial glider task assignment

- UAV/UCAV/UAS/aerial glider routing

Papers that focus on *Control Theory* for UAVs were not reviewed.

6. Critical review of the recent literature

In this section, we apply our taxonomy to 70 articles published between 2010 and 2016. We have balanced our analysis by considering articles dedicated to UAV TO/PP and UAV routing/TA. Papers devoted to technical and theoretical aspects of UAV flight dynamics were excluded from our analysis.

Table 1: Characteristics of the problems considered in this literature review.

	UAVs	
1	Multiple	A fleet of vehicles is available
2	Heterogeneous	The fleet is heterogeneous
3	Fleet size	The size of the fleet must be optimised
4	Capacity	Vehicles are capacitated
5	Geometric	The vehicle's flight dynamics are neglected
6	Dubin's	A Dubin's vehicle model has been used
7	EOM	A set of EOMs is used to model the UAVs' flight dynamics
	Waypoints	
8	Multiple	Multiple waypoints must be visited
9	Unordered	The visiting order of waypoints is unknown
10	Visits	Waypoints can be visited multiple times
11	Constraints	Special mission constraints must be considered. (e.g., time-windows and boundary conditions)
12	Covering Region	If there is a continuous covering region around the waypoints
	Environment	
13	3D	The UAVs operate in a 3D space
14	Obstacles	If obstacles are present
15	Wind	The effects of wind are considered
16	Real-time	The problem must be solved in real-time
	Launching (Depot)	
17	Multiple	There are multiple launching points
18	Inter-depot	There are inter-depots available
	Time	
19	Fixed	The UAVs' flight times between arcs are known.
20	Variable	Flight times and velocities are optimisation variables
	Approach	The type of algorithm used to solve the problem (Appendix A)
	Application	A real-world motivation to solve the problem (Appendix A)

Articles published in journals and conferences have been included in a number that we consider to be representative. Nonetheless, we apologise for any inadvertent omission of relevant papers.

The selected papers have been organised into Table 2. Each line of this table corresponds to one article and the meaning of each column relates to the numbering in Table 1. Each time an attribute is present in a paper the respective column is marked with "✕". Therefore, an empty cell indicates that its corresponding paper has not addressed the attribute indicated by this cell's column. A table with a detailed description of methods and applications for each article can be found in the Appendix A. Statistics about Table 2 are provided in Table 3.

Three types of articles can be identified in Table 2. Papers focusing on UAV routing and TA can be identified by the presence of attributes 8 and 9. The second type, which involves papers on UAV PP and TO, exclusively, correspond to the ones where attribute 9 is absent. The third type consists of articles that integrate UAV routing and PP or UAV routing and TO. The former can be identified by the presence of attributes 5 or 6 together with 8 and 9, while the latter can be identified by the presence of attributes 7, 8 and 9 together.

In Table 2 it can be seen that 70% of the articles considered a fixed flight time. This indicates that most of the UAV literature is concerned with routing and PP algorithms, where constant velocity along the trajectories is a common assumption. The EOMs of the vehicles were employed in 17.1% of the articles. In 53.8% of the papers on PP that applied a Dubin's model (which consist of only 18.6% of the total number of papers), the flight time has been considered as a variable.

Multiple UAVs were considered in 25% of the papers dealing with TO and PP. An interesting fact arises counting the number of papers dealing with multiple UAVs and their EOMs. There seems to be a preference for using PP methods and the Dubin's model when a fleet of UAVs is taken into account. One can notice that the preferred strategy is to simplify the physical models of the UAVs so as to make the problem of designing multiple flyable routes more tractable. This happens in 44.4% of the articles on UAV PP and TO and in all the articles on UAV routing and TA.

Around 37.5% of the papers on TO and PP problems dealt with visiting multiple waypoints. However, only 14.3% attempted to integrate PP and TO to routing decisions. Among them, three papers employed the UAVs' EOMs. This gives an indication that integrated routing and TO is yet to be fully investigated in the literature.

Regarding environmental conditions, 40% of the papers have studied three dimensional problems. Obstacle avoidance was tackled in 22.8% of the articles. Only a few studies (10%) included the effects of the wind in the UAVs' trajectories. In addition, only 7.1% of the papers studied real-time applications.

In 78.5% of the papers articles focusing on UAV routing and TA, a fleet of UAVs was considered. A large amount (85.7%) of the articles on routing and TA have either neglected or simplified the dynamics of the UAVs. Approximately 18% of the articles have modelled the UAVs as Dubin's vehicles. There is some overlap between these papers since some of them employ more than one methodology. This suggests the preference for simplified vehicle models when dealing with UAV routing.

Table 3 illustrates other differences between the literature on UAV TO/PP and UAV routing/TA. Each row of Table 3 shows four classes that were defined in the proposed taxonomy and their respective frequencies (defined as the number of non-empty cells divided by the total number of cells in that class).

Table 2: Summary of the taxonomic review on 70 selected papers.

Author(s)	UAVs							Waypoints					Env.				Dep.		Time	
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20
Al-Sabban et al. (2012)							×								×					×
Babel (2011)					×									×					×	
Babel (2012)					×	×								×						×
Bae et al. (2015)					×	×		×											×	
Baiocchi (2014)					×			×					×						×	
Bandeira et al. (2015)					×			×	×		×								×	
Bednowitz et al. (2012)	×				×			×	×	×									×	
Besada-Portas et al. (2010)	×				×			×					×						×	
Besada-Portas et al. (2013)	×				×			×					×						×	
Casbeer & Holsapple (2011)	×	×		×	×			×	×		×						×		×	
Chakrabarty & Langelaan (2011)				×				×					×		×					×
Chen et al. (2016)	×			×				×	×				×	×						×
Choe et al. (2016)	×			×				×					×				×		×	
Cobano et al. (2013)	×			×				×	×		×		×				×		×	
Cons et al. (2014)			×			×	×												×	
Crispin (2016)	×			×				×					×						×	
Dilão & Fonseca (2013)				×				×					×							×
Edison & Shima (2011)	×	×			×		×	×											×	
Enright et al. (2015)	×			×	×		×	×	×		×					×		×	
Evers et al. (2014)		×	×		×	×		×					×		×		×	
Faied et al. (2010)	×			×				×	×		×						×		×	
Filippis et al. (2011)	×											×						×	
Forsmo (2012)	×			×		×	×									×		×	
Fügenschuh & Müllenstedt (2015)	×	×	×	×	×	×	×		×	×	×	×		×			×		×
Furini et al. (2016)				×			×	×		×		×						×	
Gottlieb & Shima (2015)	×				×	×	×	×		×		×			×		×	
Guerriero et al. (2014)	×		×	×	×		×	×		×		×			×		×	
Han et al. (2014)		×	×									×					×	
Henchey et al. (2016)				×		×	×		×	×		×	×	×			×
Huang et al. (2016)	×			×		×						×					×	
Hu et al. (2015b)	×			×	×		×	×							×		×	
Jaishankar & Pralhad (2011)				×						×		×	×				×	
Jiang & Ng (2011)	×			×		×	×		×								×	
Kagabo (2010)				×	×					×							×	
Kivelevitch et al. (2016)	×			×		×	×								×		×	
Kumar & Padhi (2013)				×	×													×
Kwak et al. (2013)	×	×		×		×	×	×	×		×					×	
Levy et al. (2014)	×	×	×	×		×	×	×				×	×	×	×	
Liu et al. (2013)				×								×					×	
Liu et al. (2016)				×	×						×					×	
Manyam et al. (2015)	×			×	×	×	×							×		×	
Mersheeva (2015)	×		×	×	×	×	×	×						×	×	×	×	
Mufalli et al. (2012)	×		×	×	×	×										×	
Murray & Karwan (2010)	×	×	×	×	×	×	×						×		×	
Murray & Karwan (2013)	×	×	×	×	×	×	×						×		×	
Myers et al. (2016)				×	×	×		×		×		×	×		×	
Nguyen et al. (2015)				×	×	×	×		×	×						×
Niccolini et al. (2010)	×	×		×	×	×		×		×					×	
Park et al. (2012)		×	×	×	×	×	×									×	
Pepy & Hérissé (2014)				×														×
Pharpatara et al. (2015)				×					×							×	
Rogowski & Maroński (2011)				×													×	
Shanmugavel (2013)				×													×	
Silva et al. (2015)				×					×	×							×
Song et al. (2016)	×	×	×	×	×		×			×			×	×	×	
Stump & Michael (2011)	×	×	×	×	×	×	×				×		×		
Sundar & Rathinam (2014)		×	×	×	×	×	×		×		×	×	
Techy et al. (2010)				×									×			×
Thi et al. (2012)	×	×	×	×	×									×	
Vilar & Shin (2013)	×			×	×	×	×			×					×	
Wang et al. (2015)	×	×	×	×	×	×		×		×		×		×	
Wang et al. (2016)				×							×					×	
Wu et al. (2011)				×					×	×	×					×	
Xu et al. (2017)	×			×		×		×	×							×
Yakıcı (2016)	×		×	×	×	×									×	
Yang et al. (2015)	×		×	×	×	×	×		×			×		×	
Yomchinda et al. (2016)				×		×	×							×		×
Zhang et al. (2011)				×		×	×	×	×	×				×	
Zhang et al. (2012)				×	×	×	×	×	×	×					×
Zhang et al. (2014)	×			×	×	×	×	×					×		×

For example, for the articles tackling TO and PP, the number of non-empty cells for class Depot is 3 and the total number of cells for the same class is 64. Hence the density of class Depot for TO/PP papers is $3/64 = 0.047$. One can notice that while the routing/TA literature is able to include more VRP-like attributes (like multiple UAVs and waypoints), the literature on TO/PP is more concerned about modelling environmental aspects. Including environmental attributes (such as obstacles and wind) is usually possible when the UAVs physical models are integrated to the optimisation problem.

In addition, the number of articles using a fixed flight time between waypoints is higher in the routing/TA literature (89.3%) than in the TO/PP literature (59.4%). This is also related to the preference for simplified physical models in the UAV routing/TA research community.

Table 3: Densities per class for each classification.

Class	TO/PP	routing/TA
UAVs	20.1%	37.8%
Waypoints	11.9%	61.4%
Environment	26.6%	9.8%
Depot	4.7%	30.4%

The analysed papers consider a variety of different objective functions, given that they refer to different applications. Minimising the total flight time or the overall travel distance are common objectives in the UAV routing literature (e.g., Casbeer & Holsapple, 2011). In delivery applications however, minimising delivery costs or the delivery time to customers are often preferred (e.g., Wang et al., 2017). In the case of powered UAVs, minimising energy expenditure or maximising the flight duration of each UAV is a common objective (e.g., Al-Sabban et al., 2012). On the other hand, for unpowered UAVs, one usually seeks to find a trajectory that maximises the flight range (e.g., Chakrabarty & Langelaan, 2011). For UCAVs, due to the high UAV unit costs and danger involved in military missions, minimising the risk of suffering an attack is the preferred objective (e.g., Bae et al., 2015). For aerial survey operations, maximising area coverage is a popular objective (e.g., Mersheeva, 2015). In task assignment problems, service levels are usually maximised (e.g., Hu et al., 2015a). Finally, for disaster assessment and response, minimising the total mission time or the maximum flight duration (*makespan*) are amongst the most adopted objectives (e.g., Bravo & Leiras, 2015).

Figure 3 summarises the most employed algorithms for each research area described in Sections 3 and 4. These results are described in more details in the Appendix A. For the sake of simplicity, methods belonging to the same class of algorithms are grouped together on Figure 3. For example, the group *Metaheuristics* involves Iterated Local Search, Ant Colony Optimisation, Particle Swarm Optimisation, Evolutionary Algorithms, among others. Group *Heuristics* involves either methods combining more than one heuristic algorithm or specialised methods for a given problem. Group *Mixed-Integer Linear Programming* corresponds to exacts algorithms, i.e., branch-and-bound and column generation amongst others. Group *Others* on Figure 3b represents different *ad hoc* path planning algorithms that did not fit into a special category, each algorithm being present in only one paper.

One can notice that for articles involving TO (Figure 3a), exact methods for continuous optimisation are preferred. On the other hand, heuristics and metaheuristics are more frequent in the articles on UAV PP (Figure 3b). Heuristics, metaheuristics and Mixed-Integer Linear Programming (MILP) algorithms are very popular among articles considering UAV routing and TA. Being exact methods more popular in the UAV routing papers (Figure 3c) and heuristics and metaheuristics more popular in the UAV TA articles (Figure 3d).

6.1. Integrating routing and trajectory optimisation

Hereafter we highlight the contribution of articles that studied UAV routing and TO in an integrated framework. Such articles can be identified in Table 2 by attributes 7, 8 and 9 being marked with "✕".

(a) Articles on UAV trajectory optimisation.

(b) Articles on UAV path planning.

(c) Articles on UAV routing.

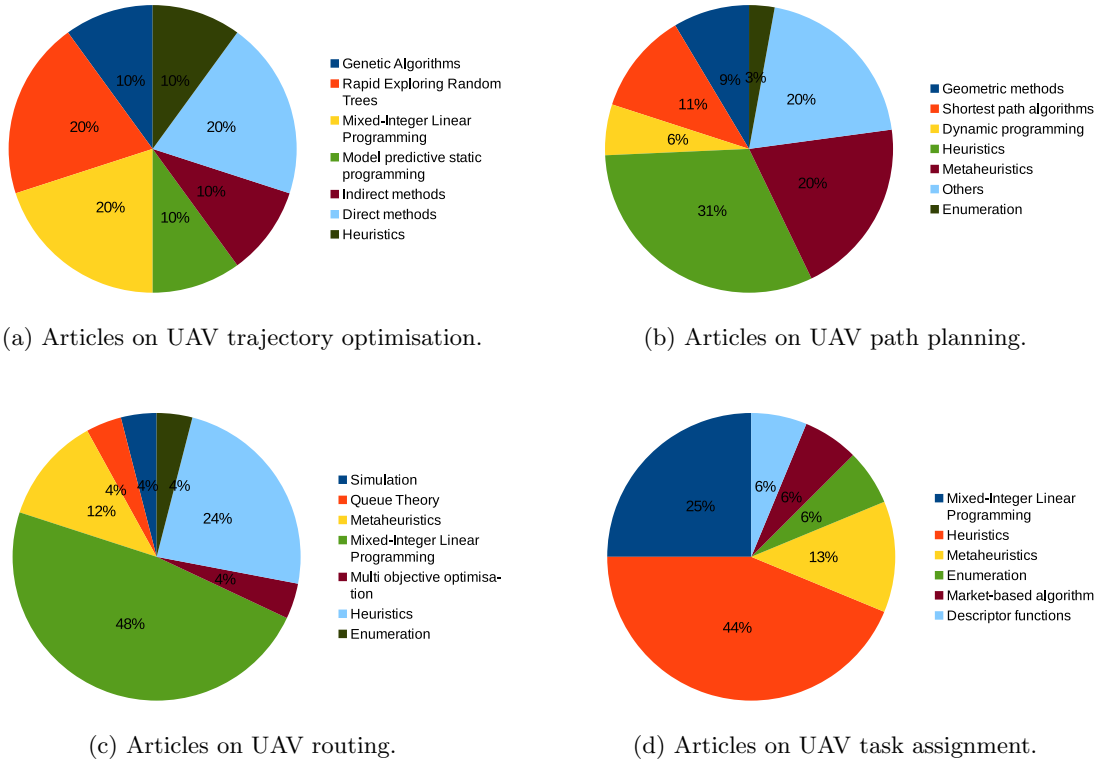(d) Articles on UAV task assignment.

Figure 3: Overview of methods and algorithms employed in the 70 selected papers.

These papers present alternative frameworks to the UAVRTOP formulation presented in Section 2.

Zhang et al. (2012) investigated the problem of routing a combat UCAV in a 3D environment through stationary ground targets whilst avoiding no-fly threat areas. In order for the attacks to succeed, the UCAV must fly within the targets' allowable attack region (which consists of a hollow-cone-like airspace around the target) and respect projectile release attitude and velocity constraints. The UCAV was modelled by high fidelity 3DOF EOMs taking wind velocities into account.

In order to solve this problem, Zhang et al. (2012) propose a hierarchical heuristic with two levels. In the first level, the vehicle's state space is discretised into a set of feasible points that intersects the targets' allowable attack region through the use of a modified probabilistic road map method. Then, for every pair of sampled points not in the same target a TO problem was solved to obtain feasible trajectories (with respect to the vehicle's dynamics and operational constraints) and their respective costs. The second decision level consists of solving a Generalised Travelling Salesman Problem (GTSP) over the network produced in the first level. This is accomplished by transforming the GTSP into an instance of the Asymmetric TSP by means of the noon-bean transformation method. The Lin-Kernighan heuristic was then employed to solve the ATSP. In addition, the authors embedded this algorithm into a real-time framework in order to make this approach more flexible for practical applications. Numerical experiments showed that this approach is computationally intensive. The authors reported that roughly 50 minutes were necessary to solve a test case with three targets and one no-fly zone.

Fügenschuh & Müllenstedt (2015) studied the problem of designing and routing a fleet of heterogeneous UAVs over a set of waypoints. The waypoints have to be selected from a list where a score was associated to each waypoint. The objective was to maximise the total score (defined as the sum of the individual scores) whilst minimising the total flight time. The UAVs' motion was modelled by piecewise

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linear dynamics based on Newton's laws of motion. The advantage of using this model lies on its simplicity, since the discretised version of these EOMs is also linear. On the other hand, the accuracy of such a model regarding UAVs flight dynamics is limited. In order to represent the range of the UAVs' sensors, the waypoints were considered to rest inside a sphere. A waypoint would be considered visited if a UAV passes through its covering sphere. No-fly zones and collision avoidance among the UAVs were also considered. Finally, different locations could be chosen to launch each UAV.

The authors proposed a Mixed-Integer Non-linear Programming (MINLP) formulation to this problem, which was linearised and could be solved by a commercial MILP optimisation software. Eight instances were created by varying the number of waypoints between 3–15, the number of no-fly zones between 0–3 and the number of UAVs between 1–2. Computational experiments showed that bigger instances with 10–15 waypoints could not be solved within one hour. The computation time required to solve smaller problems to optimality varied between 57–3400 seconds.

A similar approach was presented by Forsmo (2012). The author applied Newton's second law in order to model the motion of the UAVs. However, constraints on the magnitudes of forces, velocities and yaw rates were also imposed, which increased the complexity of the physical representation of the UAVs. Several operational constraints were considered, such as obstacles and collision avoidance. Scenarios with up to two UAVs and multiple waypoints were generated. A MILP formulation was proposed in order to find minimum flight time trajectories visiting all waypoints subject to mission and operational constraints. Computational experiments were performed over 5 test cases, constructed by varying the number of UAVs (1 or 2), waypoints (6 or 8) and by imposing, or not imposing, a visiting order. The authors showed that CPU times could be reduced by decreasing the flight time horizon.

7. Conclusions and directions for future research

The UAVRTOP is a routing problem that takes into account the flight dynamics of UAVs. UAV routing problems usually ignore flight dynamics, while work on UAV trajectory optimisation usually ignores any routing aspects. Coupling these two important aspects leads to a more realistic approach that allows the design of optimal routes and trajectories for a fleet of UAVs flying simultaneously.

This problem arises from the current development of UAV technology and the vast number of applications that these vehicles can be used for. In this paper, we first formalised the UAVRTOP. Next, an introduction to TOPs, VRPs and their variants has been provided. In addition, we introduced a taxonomy capable of classifying UAV routing/TA and UAV TO/PP problems according to their most relevant features. This taxonomy included 20 common attributes from the literature. Finally, we applied the proposed taxonomy to 70 recent papers.

The literature on UAVs routing problems has been surveyed and a lack of articles integrating UAV routing and TO has been identified. In particular, the UAVs' flight dynamics is often simplified or neglected. In many cases the behaviour of UAVs cannot be satisfactorily approximated only by their kinematics, as in the case of terrestrial robots (Forsmo, 2012). We believe that integrating the UAVs' system dynamics into routing problems is a key concept for complex operations. A realistic routing and TO algorithm must take into account the vehicle's kinematics and dynamics. In addition, by considering the UAVs' EOMs one can also better approximate, for example, the vehicles' energy consumption, which

is highly important for UAVs with limited battery duration. Modelling energy consumption is an issue that needs further investigation in the UAV routing literature.

Flight safety is an important aspect in connection with the use of UAVs. Most research on UAV routing does not consider, for example, collision avoidance and wind conditions. This is important, e.g., for goods distribution within urban areas where collisions with buildings and manned aircraft must be avoided and the fleet of UAVs must operate in a robust and reliable way.

Usually, research on UAV path and route planning concentrates on modelling kinematics. In many articles about UAV routing and TA even the kinematics are neglected. Models taking into account the forces acting on these vehicles, the interaction with wind and their manoeuvring capabilities could possibly result in computationally expensive formulations, but such models might allow for more realistic solutions.

Mathematical formulations and algorithms capable of tackling complex unmanned aerial systems in a routing framework have recently appeared in the literature. A first step in this direction has been made by Zhang et al. (2012), Forsmo (2012) and Fügenschuh & Müllenstedt (2015). Zhang et al. (2012) proposed a heuristic method based on the 3DOF EOMs of a UAV. Whereas Forsmo (2012) and Fügenschuh & Müllenstedt (2015) developed MILP formulations based on simplified dynamic equations. Only problems with limited size have been solved by the aforementioned authors. Therefore, the development of efficient frameworks for solving UAVRTOPs still raises challenging research questions that need to be answered.

Appendix A. Methods and applications for the selected literature

Table A.4 highlights the methods and practical applications addressed in the 70 selected papers.

Table A.4: Summary of methods and applications on 70 selected papers.

Authors	Approach	Application
Al-Sabban et al. (2012)	Markov decision process	Path planning in uncertain wind conditions
Babel (2011)	Shortest path algorithms	UAV path planning with obstacles
Babel (2012)	Shortest path algorithms	Path planning in a risk environment
Bae et al. (2015)	Dynamic programming and heuristics	Risk-constrained shortest path for UCAV
Baiocchi (2014)	Heuristic algorithms	Path planning for aerial photography
Bandeira et al. (2015)	Heuristic algorithms	UAV routing for aerial photography
Bednowitz et al. (2012)	Simulation model	UAV routing in dynamic environment
Besada-Portas et al. (2010)	Evolutionary algorithms	Real-time UAV path planning
Besada-Portas et al. (2013)	Evolutionary algorithms	Real-time UAV path planning
Casbeer & Holsapple (2011)	Column generation	UAV TA with precedence
Chakrabarty & Langelaan (2011)	Energy map method	Path planning for soaring UAVs
Chen et al. (2016)	Genetic algorithm	Multi UAV trajectory optimisation
Choe et al. (2016)	Pythagorean hodograph bézier curves	Cooperative path planning
Cobano et al. (2013)	Rapid exploring random trees	Cooperative trajectory optimisation
Cons et al. (2014)	Heuristic algorithms	Integrated TA and path planning
Crispin (2016)	Rapid exploring random trees	Path planning for aerial gliders
Dilão & Fonseca (2013)	Heuristic algorithms	Path planning for a hypersonic glider
Edison & Shima (2011)	Genetic algorithm	Integrated TA and path planning
Enright et al. (2015)	Queueing theory	UAV routing in stochastic environments
Evers et al. (2014)	ILS metaheuristic	UAV orienteering problem with time windows
Faied et al. (2010)	Mixed-Integer Linear Programming	Multi UAV routing problem
Filippis et al. (2011)	Shortest path algorithms	UAV path planning with obstacles
Forsmo (2012)	Mixed-Integer Linear Programming	UAV routing and trajectory optimisation
Fügenschuh & Müllenstedt (2015)	Mixed-Integer Linear Programming	UAV routing and trajectory optimisation
Furini et al. (2016)	Mixed-Integer Linear Programming	Time dependent UAV routing problem
Gottlieb & Shima (2015)	Enumerative and heuristic algorithms	Integrated TA and path planning
Guerriero et al. (2014)	Multi-objective optimisation	UAV routing with time windows
Han et al. (2014)	Dynamic programming	UAV path planning
Henchey et al. (2016)	Enumerative and heuristic algorithms	UAV routing problems
Huang et al. (2016)	Ant colony optimisation	Multi UAV path planning
Hu et al. (2015b)	Ant colony optimisation	UAV task assignment
Jaishankar & Pralhad (2011)	Multi criteria decision analysis	UAV path planning
Jiang & Ng (2011)	Mixed-Integer Linear Programming	Multi UAV routing problem
Kagabo (2010)	Fuzzy Logic	Path planning for aerial gliders
Kivelevitch et al. (2016)	Market-based algorithm	UAVs TA problem
Kumar & Padhi (2013)	Model predictive static programming	UAV trajectory optimisation

Table A.4 – continued from previous page

Authors	Approach	Application
Kwak et al. (2013)	Heuristic algorithms	Generalised UAVs TA
Levy et al. (2014)	Heuristic algorithms	UAVs routing with refuelling depots
Liu et al. (2013)	Heuristic algorithms	Real-time UAV path planning
Liu et al. (2016)	Collocation interval analysis method	UAV path planning
Manyam et al. (2015)	Lagrangian Relaxation	Multi depot UAVs routing
Mersheeva (2015)	Heuristics and constraint programming	UAV routing in disaster assessment
Mufalli et al. (2012)	Mixed-Integer Linear Programming and heuristics	UAVs routing problems
Murray & Karwan (2010)	Mixed-Integer Linear Programming	UAVs dynamic TA and routing
Murray & Karwan (2013)	Branch-and-bound	UAVs dynamic routing
Myers et al. (2016)	Shortest path algorithm	Real-time UAV path planning
Nguyen et al. (2015)	Look-up tables	Path planning for aerial gliders
Niccolini et al. (2010)	Descriptor functions methodology	Multi UAV TA problem
Park et al. (2012)	Heuristic algorithms	UAV routing
Pepy & Hérissé (2014)	Indirect shooting method	Trajectory optimisation for an aerial glider
Pharpatara et al. (2015)	Geometric path planning	Path planning for a hypersonic glider
Rogowski & Maroński (2011)	Direct pseudospectral method	Trajectory optimisation for an aerial glider
Shanmugavel (2013)	Bayesian rule-based algorithm	UAV path planning
Silva et al. (2015)	Non-linear programming	Trajectory optimisation for an aerial glider
Song et al. (2016)	Mixed-Integer Linear Programming and heuristics	Multi UAV TA problem
Stump & Michael (2011)	Mixed-Integer Linear Programming	Multi UAV routing problem
Sundar & Rathinam (2014)	Heuristic and approximation algorithms	UAV routing with refueling depots
Techy et al. (2010)	Heuristic algorithm	UAV path planning in uniform wind
Thi et al. (2012)	Exact and heuristic algorithms	UAVs task assignment
Vilar & Shin (2013)	Heuristic algorithm	Communication-aware TA problem
Wang et al. (2015)	Heuristic algorithms	Multi UAV TA problem
Wang et al. (2016)	Population-based algorithms	UAV path planning
Wu et al. (2011)	Genetic Algorithm	UAV path planning
Xu et al. (2017)	Gradient-descent algorithm	UAV path planning
Yakıcı (2016)	Ant Colony Optimisation	UAV location and routing problem
Yang et al. (2015)	Heuristic algorithms	UAV path planning
Yomchinda et al. (2016)	Parametrization techniques and heuristics	Aircraft path planning
Zhang et al. (2011)	Differential Evolution Algorithm	UAVs real-time path planning
Zhang et al. (2012)	Heuristic algorithm	UAV routing and trajectory optimisation
Zhang et al. (2014)	Memetic Algorithm	UAV routing problem

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